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APPLICATION OF GRADIENT PROJECTION METHOD TO PARAMETRIC OPTIMIZATION OF STEEL LATTICE PORTAL FRAME

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Abstract. The paper has proposed a mathematical model for parametric optimization problem of the steel lattice portal frame. The design variable vector includes geometrical parameters of the structure (node coordinates), as well as cross-sectional dimensions of the structural members. The system of constraints covers load-carrying capacities constraints formulated for all design sections of structural members of the steel structure subjected to all ultimate load case combinations. The displacements constraints formulated for the specified nodes of the steel structure subjected to all serviceability load case combinations have been also included into the system of constraints. Additional requirements in the form of constraints on lower and upper values of the design variables, constraints on permissible minimal thicknesses, constraints on permissible maximum diameter-to-thickness ratio for the structural members with circle hollow sections, as well as the conditions for designing gusset-less welded joints between structural members with circle hollow sections have been also considered in the scope of the mathematical model. The method of the objective function gradient projection onto the active constraints surface with simultaneous correction of the constraints violations has been used to solve the formulated parametric optimization problem. New optimal layouts of the steel lattice portal frame by the criterion of the minimum weight, as well as minimum costs on manufacturing and erection have been presented.

Keywords: optimization, steel lattice frame, nonlinear programming, strength, buckling, stiffness, gradient projection method, finite element method, numerical algorithm

ПРИМЕНЕНИЕ МЕТОДА ПРОЕКЦИИ ГРАДИЕНТА ДЛЯ ПАРАМЕТРИЧЕСКОЙ ОПТИМИЗАЦИИ СТАЛЬНОЙ РЕШЕТЧАСТОЙ РАМЫ

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Аннотация. В статье предложена математическая модель для задачи параметрической оптимизации стальной решетчатой поперечной рамы каркаса здания, несущие элементы которой выполнены из круглых труб. Вектор переменных проектирования содержит геометрические параметры конструкции (координаты узлов), а также размеры поперечных сечений несущих элементов конструкции. Система ограничений включает ограничения несущей способности, сформулированные для всех расчетных сечений элементов конструкции, подлежащей действию всех комбинаций нагрузок первой группы предельных состояний. В систему ограничений также включены ограничения перемещений узлов, сформулированные для определенных узлов конструкции, подлежащей действию всех комбинаций нагрузок второй группы предельных состояний. Дополнительные ограничения в форме ограничений на верхнюю и нижнюю границы варьирования переменных проектирования, ограничения на допустимую минимальную толщину сечения, ограничения на допустимое максимальное отношение диаметра к толщине трубы, а также условия конструирования бесфасоночных узлов решетчатой конструкции с элементами их круглых труб также были рассмотрены в составе системы ограничений математической модели. Для решения сформулированной задачи параметрической оптимизации использовался метод проекции градиента функции цели на поверхность активных ограничений при одновременной ликвидации невязок в нарушенных

ограничениях. В результате получены новые оптимальные проектные решения стальной решетчатой поперечной рамы по критерию минимума массы конструкции, а также по критерию минимума сметной стоимости на ее изготовление и возведение.

Ключевые слова: оптимизация, стальная решетчатая рама, нелинейное программирование, прочность, устойчивость, жесткость, градиентный метод, метод конечных элементов, численный алгоритм.

INTRODUCTION

Over the past 50 years, numerical optimization and the finite element method have individually made significant advances and have together been developed to make possible the emergence of structural optimization as a potential design tool. In recent years, great efforts have been also devoted to integrate optimization procedures into the CAD facilities. With these new developments, lots of computer packages are now able to solve relatively complicated industrial design problems using different structural optimization techniques.

Applied optimum design problems for bar structures in some cases are formulated as parametric optimization problems, namely as searching problems for unknown structural parameters, which provide an extreme value of the specified purpose function in the feasible region defined by the specified constraints [1]. In this case, structural optimization is performed by variation of the structural parameters when the structural topology, cross-section types and node type connections of the bars, the support conditions of the bar system, as well as loading patterns and load design values are prescribed and constants.

Kibkalo et al. in the paper [2] formulated a parametric optimization problem for thin-walled bar structures and considered methods to solve them. The searching for the optimum solution has been performed by varying the structural parameters providing the required load-carrying capacity of structural members and the minimum value of manufacturing costs.

Alekseytsev has described the process of developing a parametrical-optimization algorithm for steel trusses in the paper [3].

Parametric optimization has been performed taking into account strength, stability and stiffness constraints formulated for all truss members.

Serpik et al. in the paper [4] developed an algorithm for parametric optimization of steel flat rod systems. The optimization problem has been formulated as a structural weight minimization problem taking into account strength and displacement constraints, as well as overall stability constraints. The cross-sectional dimensions of the truss members and the coordinates of the truss panel joints have been considered as design variables. The structural analysis of internal forces and displacements for considered structures has been performed using the finite element method. An iterative procedure for searching for optimum solution has been proposed in [5].

Sergeyev et al. in the paper [6] formulated a parametric optimization problem with constraints on faultless operation probability of bar structures with random defects. The weight of the bar structures has been considered as the objective function. Initial global imperfections have been considered as small independent random variables distributed according to normal distribution law, as well as buckling load value has been also considered as a random variable.

The mathematical model of the parametric optimization problem of structures includes a set of design variables, an objective function, as well as constraints, which reflect generally non-linear dependences between them [7]. If the purpose function and constraints of the mathematical model are continuously differentiable functions, as well as the search space is smooth, then the parametric

optimization problems are successfully solved using gradient projection non-linear methods [8]. The gradient projection methods operate with the first derivatives or gradients only both of the objective function and constraints. The methods are based on the iterative construction of such a sequence of the approximations of design variables that provides convergence to the optimum solution (optimum values of the structural parameters) [9].

Additionally, a sensitivity analysis is a useful optional feature that could be used in scope of the numerical algorithms developed based on the gradients methods [10]. Thus, in the paper [11] Sergeyev et al. formulated a parametric optimization problem of linearly elastic space frame structures taking into account the stress and multiple natural frequency constraints. The cross-sectional parameters of structural members as well as node positions of the considered structures has been considered as design variables. The sensitivity analysis of multiple frequencies has been performed using analytic differentiation with respect to the design variables. The optimal design of the structure has been obtained by solving a programming sequence of quadratic problems.

In this paper, steel lattice portal frame is considered as research object, investigated for the searching for optimum parameters of the structural form. The following research tasks are formulated: to develop a mathematical model for parametric optimization of the considered steel structures taking into account load-carrying capacities and stiffness constraints; to propose a numerical algorithm for parametric optimization of the steel structures based on the gradient projection method; to confirm the validity of the optimum solutions obtained using the proposed methodology based on numerical examples.

1. PROBLEM FORMULATION FOR PARAMETRIC OPTIMIZATION OF STEEL STRUCTURES

Let us consider a parametric optimization problem of a structure consisting of bar members. The problem statement can be performed taking into account the following assumptions widely used in structural mechanic problems: the material of the structure is ideal elastic; the bar structure is deformable linearly; external loadings applied to the structure are quasi-static.

Let us also formulate the following preconditions for calculation: cross-section types and dimensions of structural members are constant along member lengths; external loadings are applied to the structural members without eccentricities relating to the center of mass and shear center of its cross-sections; an additional restraining by stiffeners are provided in the design sections where point loads (reactions) applied with the exception of crosssection warping and local buckling of the crosssection elements; load-carrying capacity of the structural joints, splices and connections are provided by additional structural parameters do not covered by the considered parametric optimization problem.

A parametric optimization problem of the structure can be formulated as presented below: to find optimum values for geometrical parameters of the structure, member's crosssection dimensions and initial pre-stressing forces introduced into the specified redundant members of the bar system, which provide the extreme value of the determined optimality criterion and satisfy all load-carrying capacities and stiffness requirements. We assume, that the structural topology, cross-section types and node type connections of the bars, the support conditions of the bar system, as well as loading and pre-stressing patterns are prescribed and constants.

The formulated parametric optimization problem can be considered integrally using the mathematical model in the form of the nonlinear programming task including an objective function, a set of independent design variables and constraints, which reflect generally nonlinear dependences between them. The validity of the mathematical model can be estimated by the compliance of its structure with the design code requirements.

The parametric optimization problem of steel structures can be stated in the following mathematical terms: to find unknown structural parameters $\vec{X} = \left\{X_{\iota}\right\}^{T}$, $\iota = \overline{1, N_{X}}$ (N_{X} is the total number of the design variables), providing the least value of the determined objective function:

$$f^* = f(\vec{X}^*) = \min_{\vec{X} \in \mathfrak{I}_{\square}} f(\vec{X}), \qquad (1.1)$$

in a feasible region (search space) \Im defined by the following system of constraints:

$$\psi(\vec{X}) = \left\{ \psi_{\kappa}(\vec{X}) = 0 \mid \kappa = \overline{1, N_{EC}} \right\}; \tag{1.2}$$

$$\mathbf{\varphi}\left(\vec{X}\right) = \left\{\phi_{\eta}\left(\vec{X}\right) \le 0 \mid \eta = \overline{N_{EC} + 1, N_{IC}}\right\}; \qquad (1.3)$$

where \vec{X} is the vector of the design variables (unknown structural parameters); f, ψ_{κ} , φ_{η} are the continuous functions of the vector argument; \vec{X}^* is the optimum solution or optimum point (the vector of optimum values of the structural parameters); f^* is the optimum value of the optimum criterion (objective function); N_{EC} is the number of constraints-equalities $\psi_{\kappa}(\vec{X})$, which define hyperplanes of the feasible solutions; N_{IC} is the number of constraints-inequalities $\varphi_{\eta}(\vec{X})$, which define a feasible region in the design space \Im .

The vector of the design variables comprises of unknown geometrical parameters of the structure $\vec{X}_G = \left\{X_{G,\chi}\right\}^T$, $\chi = \overline{1,N_{X,G}}$, and unknown cross-sectional dimensions of the

structural members
$$\vec{X}_{CS} = \left\{ X_{CS,\alpha} \right\}^T$$
, $\alpha = \overline{1, N_{X,CS}}$:

$$\vec{X} = \left\{ \vec{X}_G, \vec{X}_{CS} \right\}^T = \left\{ \left\{ X_{G,\chi} \right\}, \left\{ X_{CS,\alpha} \right\} \right\}^T;$$
 (1.4)

where $N_{X,G}$ is the total number of unknown node coordinates of the steel structure; $N_{X,CS}$ is the total number of unknown cross-sectional dimensions of the structural members, $N_{X,G} + N_{X,CS} = N_X$.

The specific technical-and-economic index (material weight, material cost, construction cost etc.) or another determined indicator can be considered as the objective function Eq. (1.1) taking into account the ability to formulate its analytical expression as a function of design variables \vec{X} .

Load-carrying capacities constraints (strength and stability inequalities) formulated based on the design code requirements [12] for all design sections of the structural members subjected to all design load combinations at the ultimate limit state as well as displacements constraints (stiffness inequalities) for the specified nodes of the bar system subjected to all design load combinations at the serviceability limit state should be included into the system of constraints Eqs. (1.2) - (1.3). Additional requirements, which describe structural, technological and serviceability particularities of the considered structure can be also included into the system Eqs. (1.2) - (1.3).

The design internal forces in the structural members used in the strength and stability inequalities of the system Eqs. (1.2) - (1.3) are considered as state variables depending on design variables \vec{X} and can be calculated from the following linear equations system of the finite element method [13], $k = 1, N_{LC}^{ULS}$:

$$\mathbf{K}(\vec{X}_G, \vec{X}_{CS}) \times \vec{z}_{ULS,k} = \vec{p}_{ULS,k}(\vec{X}_G); \tag{1.5}$$

where $\mathbf{K}(\vec{X}_G, \vec{X}_{CS})$ is the stiffness matrix of the finite element model of the bar system, which should be formed depending on the unknown (variable) cross-sectional dimensions of the structural members \vec{X}_{CS} , as well as unknown (variable) node coordinates of the structure \vec{X}_G ; $\vec{p}_{ULS,k}(\vec{X}_G)$ is the column-vector of the node's loads for k th design load combination of the ultimate limit state, which should be formed depending on unknown (variable) coordinates of the structure \vec{X}_G ; $\vec{z}_{ULS,k}$ is the result column-vector of the node displacements for k th design load combination of the ultimate limit state, $\vec{z}_{ULS,k} = \mathbf{Z}_{\text{FEM},k}^{ULS} \left(\vec{X}_G, \vec{X}_{CS} \right) = \mathbf{Z}_{\text{FEM},k}^{ULS} \left(\vec{X} \right);$ N_{LC}^{ULS} is the number of the design ultimate load combinations. For each ith design section of j th structural member subjected to k th ultimate design load combination the design internal forces (axial force, bending moments and shear forces) can be calculated depending on node displacement column-vector $\vec{z}_{ULS.k}$.

The node displacement of the bar system used in stiffness inequalities of the system Eqs. (1.2) - (1.3) are also considered as state variables depending on design variables \vec{X} and can be calculated from the following linear equations system of the finite element method [13], $k = 1, N_{LC}^{SLS}$:

$$\mathbf{K}(\vec{X}_G, \vec{X}_{CS}) \times \vec{z}_{SLS,k} = \vec{p}_{SLS,k}(\vec{X}_G); \quad (1.6)$$

where $\vec{p}_{\mathit{SLS},k}\left(\vec{X}_{\mathit{G}}\right)$ is the column-vector of the node's loads for k th design load combination of the serviceability limit state, which should be formed depending on unknown (variable) node coordinates of the structure \vec{X}_G ; $\vec{z}_{ULS,k}$ is column-vector result of the node displacements for k th design load combination of serviceability the limit state,

$$\vec{z}_{SLS,k} = \mathbf{Z}_{\text{FEM},k}^{SLS} \left(\vec{X}_G, \vec{X}_{CS} \right) = \mathbf{Z}_{\text{FEM},k}^{SLS} \left(\vec{X} \right); \ N_{LC}^{SLS}$$
 is the number of the design serviceability load

the number of the design serviceability load combinations. For each m th node of the finite element model subjected to k th serviceability design load combination the design vertical and horizontal displacements can be calculated depending on node displacement column-vector $\vec{z}_{SLS,k}$.

2. IMPROVED GRADIENT PROJECTION METHOD TO SOLVE THE FORMULATED PARAMETRIC OPTIMIZATION PROBLEM

The parametric optimization problem stated as non-linear programming task by Eqs. (1.1) – (1.3) can be solved using a gradient projection method. The method of objective function gradient projection onto the active constraints surface with simultaneous correction of the constraints violations ensures effective searching for solution of the non-linear programming tasks occurred when optimum designing of the building structures [14, 15]. The gradient projection method operates with the first derivatives or gradients only of both the objective function Eq. (1.1) and constraints Eqs. (1.2) - (1.3). The method is based on the iterative construction of such sequence Eq. (2.1) of the approximations of the design variables $\vec{X} = \{X_t\}^T$, $t = \overline{1, N_X}$, that provides convergence to the optimum solution (optimum values of the structural parameters):

$$\vec{X}_{t+1} = \vec{X}_t + \Delta \vec{X}_t \,, \tag{2.1}$$

where $\vec{X}_t = \left\{X_t\right\}^T$, $t = \overline{1, N_X}$ is the current approximation to the optimum solution \vec{X}^* that satisfies both constraints-equalities Eq. (1.2) and constraints-inequalities Eq. (1.3) with the extreme value of the objective function Eq. (1.1); $\Delta \vec{X}_t = \left\{\Delta X_t\right\}^T$, $t = \overline{1, N_X}$, is the increment vector for the current values of the

design variables \vec{X}_t ; t is the iteration's index. The start point of the iterative searching process $\vec{X}_{t=0}$ can be assigned as engineering estimation of the admissible design of the structure.

The active constraints only of constraints system Eqs. (1.2) - (1.3) should be considered at each iteration. A set of active constraints numbers \mathbf{A} calculated for the current approximation \vec{X}_t to the optimum solution (current design of the structure) is determined as:

$$\mathbf{A} = \mathbf{\kappa} \cup \mathbf{\eta},$$

$$\mathbf{\kappa} = \left\{ \kappa \middle| \left| \psi_{\kappa} \left(\vec{X}_{t} \right) \right| \ge -\varepsilon \right\},$$

$$\mathbf{\eta} = \left\{ N_{EC} + \eta \middle| \phi_{\eta} \left(\vec{X}_{t} \right) \ge -\varepsilon \right\},$$
(2.2)

where ε is a small positive number introduced here in order to diminish the oscillations on movement alongside of the active constraints surface.

The increment vector $\Delta \vec{X}_t$ for the current values of the design variables \vec{X}_t can be determined by the following equation:

$$\Delta \vec{X}_t = \Delta \vec{X}_{\perp}^t + \Delta \vec{X}_{\square}^t, \qquad (2.3)$$

where $\Delta \vec{X}_{\perp}^t$ is the vector calculated subject to the condition of elimination the constraint's violations; $\Delta \vec{X}_{\square}^t$ is the vector determined taking into consideration the improvement of the objective function value. Vectors $\Delta \vec{X}_{\square}^t$ and $\Delta \vec{X}_{\perp}^t$ are directed parallel and perpendicularly accordingly to the subspace with the vectors basis of the linear-independent constraint's gradients, such that:

$$\left(\Delta \vec{X}_{\perp}^{t}\right)^{T} \Delta \vec{X}_{\square}^{t} = 0. \tag{2.4}$$

The values of the constraint's violations for the current approximation \vec{X}_t of the design

variables are accumulated into the following vector:

$$\mathbf{V} = \left(\psi_{\kappa} \left(\vec{X} \right) \forall \, \kappa \in \mathbf{K}; \ \phi_{\eta} \left(\vec{X} \right) \forall \, \eta \in \mathbf{\eta} \right).$$

Let us introduce a set L, $L \subseteq A$, of the constraint's numbers, such that the gradients of the constraints at the current approximation \vec{X}_t to the optimum solution are linear-independent. Component $\Delta \vec{X}_{\perp}^t$ is calculated from the equation presented below:

$$\Delta \vec{X}_{\perp}^{t} = \left[\nabla \varphi \right] \vec{\mu}_{\perp}, \qquad (2.5)$$

where $\left[\nabla\varphi\right]$ is the matrix that consists of components $\frac{\partial\psi_{\kappa}}{\partial X_{\iota}}$ and $\frac{\partial\phi_{\eta}}{\partial X_{\iota}}$, here $\iota=\overline{1,N_{\chi}}$,

 $\kappa \in \mathbf{L}$, $\eta \in \mathbf{L}$; $\vec{\mu}_{\perp}$ is the column-vector that defines the design variables increment subject to the condition of elimination the constraint's violations. Vector $\vec{\mu}_{\perp}$ can be calculated as presented below.

In order to correct constraint's violations V, vector $\Delta \vec{X}_{\perp}^t$ to a first approximation should also satisfy Taylor's theorem for the continuously differentiable multivariable function in the vicinity of point \vec{X}_t for each constraint from set L, namely:

$$-\mathbf{V} = \left[\nabla \varphi\right]^T \Delta \vec{X}_{\perp}^t. \tag{2.6}$$

With substitution of Eq. (2.5) into Eq. (2.6) we obtain the system of equations to determine column-vector $\vec{\mu}_{\perp}$:

$$\left[\nabla\varphi\right]^{T}\left[\nabla\varphi\right]\vec{\mu}_{\perp} = -\mathbf{V}. \tag{2.7}$$

Component $\Delta \vec{X}_{\square}^t$ is determined using the following equation:

$$\Delta \vec{X}_{\square}^{t} = \xi \times \vec{p}_{\nabla f} = \xi \left(\nabla \vec{f} - \left[\nabla \varphi \right] \vec{\mu}_{\square} \right), \tag{2.8}$$

where $\nabla \vec{f}$ is the vector of the objective function gradient in the current point (current approximation of the design variables) \vec{X}_t ; $\vec{p}_{\nabla f}$ is the projection of the objective function gradient vector onto the active constraints surface in the current point \vec{X}_t ; $\vec{\mu}_{\square}$ is the column-vector that defines the design variable's increment subject to the improvement of the objective function value. Column-vector $\vec{\mu}_{\square}$ can be calculated approximately using the least-square method by the following equation:

$$\left[\nabla\varphi\right]\vec{\mu}_{\square}\approx\nabla\vec{f}\;,$$
 (2.9)

or from the equation presented below:

$$\left[\nabla\varphi\right]^{T}\left[\nabla\varphi\right]\vec{\mu}_{\square} = \left[\nabla\varphi\right]^{T}\nabla\vec{f}; \qquad (2.10)$$

where ξ is the step parameter, which can be calculated subject to the desired increment Δf of the purpose function on movement along the direction of the purpose function anti-gradient. The increment Δf can be assign as 5...25% from the current value of the objective function $f(\vec{X}_t)$:

$$\Delta f = \xi \left(\nabla \vec{f} \right)^T \nabla \vec{f} , \ \xi = \frac{\Delta f}{\left(\nabla \vec{f} \right)^T \nabla \vec{f}},$$
 (2.11)

where in case of minimization Eq. (1.1) Δf and ξ accordingly have negative values. The parameter ξ can be also calculated using the dependency presented below:

$$\xi = \frac{\Delta f}{\left(\vec{p}_{\nabla f}\right)^T \nabla \vec{f}}, \qquad (2.12)$$

that follows from the condition of attainment the desired increment of the objective function Δf

on the movement along the direction of the objective function anti-gradient projection onto the active constraints surface. Step parameter ξ can be also selected as a result of numerical experiments performed for each type of the structure individually [16, 17].

Using Eqs. (2.5) and (2.8), Eq. (2.3) can be rewritten as presented below:

$$\Delta \vec{X}_{t} = \left[\nabla \varphi\right] \vec{\mu}_{\perp} + \xi \left(\nabla \vec{f} - \left[\nabla \varphi\right] \vec{\mu}_{\square}\right), \quad (2.13)$$

Or

$$\Delta \vec{X}_t = \xi \ \nabla \vec{f} + \left[\nabla \varphi \right] \left(\vec{\mu}_{\perp} - \xi \ \vec{\mu}_{\square} \right), \quad (2.14)$$

where column-vectors $\vec{\mu}_{\perp}$ and $\vec{\mu}_{\square}$ are calculated using Eq. (2.7) and Eq. (2.9) or Eq. (2.10), respectively.

The linear-independent constraints of the system Eqs. (1.2) - (1.3) should be detected when constructing the matrix of the active constraints gradients $[\nabla \varphi]$ used by Eq. (2.7) and Eq. (2.9) Selection of Eq. (2.10). the independent constraints can be performed based on the equivalent transformations of the resolving equations of the gradient projection method using the non-degenerate transformation matrix H, such that the sub-diagonal elements of the matrix $\mathbf{H}[\nabla \varphi]$ equal to zero. An orthogonal matrix of the elementary mapping (Householder's transformation) [18] has been used to select linear-independent constraints of the system Eqs. (1.2) - (1.3) as well as to form triangular structure of the nonzero elements of matrix $\mathbf{H}[\nabla \varphi]$ [14].

Using Householder's transformations described above triangular structure of the nonzero elements of matrix $\mathbf{H}[\nabla \varphi]$ is formed step-by-step. Besides, Eq. (2.7) and Eq. (2.9) can be rewritten as follow:

$$([\nabla \varphi]^T \mathbf{H}^T)(\mathbf{H}[\nabla \varphi]) \vec{\mu}_{\perp} = -\mathbf{V}; \qquad (2.15)$$

$$\mathbf{H} \left[\nabla \varphi \right] \vec{\mu}_{\square} \approx \mathbf{H} \nabla \vec{f} \ . \tag{2.16}$$

Equivalent Householder transformations of the resolving equations Eqs. (2.15), (2.16) have been proposed by the paper [14]. They increase numerical efficiency of the algorithm developed based on the considered method.

In order to calculate column-vectors $\vec{\mu}_{\perp}$ and $\vec{\mu}_{\square}$, it is required only to perform forward and backward substitutions in Eq. (2.15) and Eq. (2.16).

To accelerate the convergence of the minimization algorithm presented above, h th columns should be excluded from matrix $\mathbf{H}[\nabla \varphi]$. These columns correspond to those constraints from Eq. (1.3), for which the following inequality satisfies:

$$\mu_{\perp h} - \xi \times \mu_{\square h} > 0. \tag{2.17}$$

As presented by the papers [14, 15], when $\mu_{\perp h} - \xi \times \mu_{\square h} > 0$, then the return onto the active constraints surface from the feasible region \Im is performed with simultaneous degradation of the objective function value. At the same time, in case of $\mu_{\perp h} - \xi \times \mu_{\square h} < 0$, both the improvement of the objective function value and the return from the inadmissible region onto the active constraints surface are performed.

When excluding h th columns from matrix $\mathbf{H}[\nabla \varphi]$ corresponded to those constraints for which Eq. (2.17) is satisfied, the matrix $(\mathbf{H}[\nabla \varphi])_{red}$ with a broken (non-triangular) structure of the non-zero elements is obtained. The set L of the linear-independent active constraints numbers transforms into the set L_{red} respectively. At the same time, the vector of the constraint's violations V reduced into the vector V_{red} accordingly. In order to restore the triangular structure of the matrix $(\mathbf{H}[\nabla \varphi])_{red}$ with zero sub-Givens transformations diagonal elements, (Givens rotations) [18] can be used.

Considering Givens transformations, Eq. (2.15) and Eq. (2.16) for column-vectors $(\vec{\mu}_{\perp})_{red}$ and $(\vec{\mu}_{\parallel})_{red}$ can be rewritten as:

$$([\nabla \varphi]^T \mathbf{H}^T)_{red} \mathbf{G}^T \times \times \mathbf{G} (\mathbf{H} [\nabla \varphi])_{red} (\vec{\mu}_{\perp})_{red} = -\mathbf{V}_{red};$$

$$(2.18)$$

$$\mathbf{G} (\mathbf{H} [\nabla \varphi])_{red} (\vec{\mu}_{\parallel})_{red} \approx \mathbf{G} \mathbf{H} \nabla \vec{f}.$$

$$(2.19)$$

Equivalent transformations of the resolving equations Eqs. (2.18), (2.19) using Givens rotations (transformations with matrix **G**) ensure acceleration of the iterative searching process Eq. (2.1) in those cases when Eq. (2.17) takes into account due to decreasing the amount of calculations [14].

The main resolving equation of the gradient method Eq. (2.13) and Eq. (2.14) can be rewritten as presented below:

$$\Delta \vec{X}_{t} = \left(\mathbf{H} \left[\nabla \varphi\right]\right)_{red} \left(\vec{\mu}_{\perp}\right)_{red} + \xi \left(\nabla \vec{f} - \left(\mathbf{H} \left[\nabla \varphi\right]\right)_{red} \left(\vec{\mu}_{\square}\right)_{red}\right);$$
(2.20)

Or

$$\begin{split} &\Delta \vec{X}_{t} = \xi \ \nabla \vec{f} \ + \\ &+ \left(\mathbf{H} \big[\nabla \varphi \big] \right)_{red} \left(\left(\vec{\mu}_{\perp} \right)_{red} - \xi \left(\vec{\mu}_{\square} \right)_{red} \right). \end{split} \tag{2.21}$$

It should be noted that the lengths of the gradient vectors for the objective function Eq. (1.1), as well as for constraints Eqs. (1.2) - (1.3), remain as they were in scope of the proposed equivalent transformations ensuring the dependability of the optimization algorithm [14].

The determination the convergence criterion is the final question when using the iterative searching for the optimum point Eq. (2.1) described above. Considering the geometrical content of the gradient steepest descent method, we can assume that at the permissible point \vec{X}_t the component of the increment vector $\Delta \vec{X}_{\perp}^t$ for the design variables

should be vanish, $\Delta \vec{X}_{\square}^{t} \rightarrow 0$, in case of approximation to the optimum solution of the nonlinear programming task presented by Eqs. (1.1) – (1.5). So, the following convergence criterion of the iterative procedure Eq. (2.1) can be assigned:

$$\left\| \Delta \vec{X}_{\square}^{k} \right\| = \sqrt{\sum_{t=1}^{N_{X}} \left(\Delta X_{\square,t}^{k} \right)^{2}} < \varepsilon_{1}, \qquad (2.22)$$

where ε_1 is a small positive number. In the paper [14] the convergence criteria for the iterative procedure Eq. (2.1) has been presented in detail.

3. MATHEMATICAL MODEL FOR PARAMETRIC OPTIMIZATION OF THE LATTICE PORTAL FRAME

A parametric optimization task for lattice portal frame of the steel warehouse framework designed as repository for the granulated sulfur has been considered. Building object locates in seaport Ust-Luga of Russian Federation The general building sizes are length 247.25 m and width 69.0 m. Steel framework of the building

consists of portal frames with span 69.0 m positioned along building length with bay 7.5 m. Steel portal frames consist of the lattice structural members fabricated from pipes with steel grade St20 according to design code [12]. Joints in the lattice structural members were designed using welded connections without gussets.

There is a service platform at the level +28.25 m provided for supporting the crane-loader and conveyor (see Fig. 3.1). The load-bearing structures of the platform are suspended to the structural members of the portal frames. Welded I-beams of this platform were manufactured from the universal steel sheets of grade S245.

Design scheme of the steel lattice portal frame was assumed as a hinged-bar structure with hinged column bases. Geometrical scheme of the portal frame was described using the set of nodes and bars with orientation on implementation of the finite element method for linear static analysis. Node coordinates of the design scheme were determined in Cartesian coordinate system and presented as expressions in dependence of geometrical design variables of the optimization task.



<u>Figure 3.1.</u> Assembling process on job site (Photo has been provided by V. Shymanovsky Ukrainian Research and Design Institute of Steel Construction)

Dead loads included self weight of the portal frame, roof purlins, roof bracings, fire escape staircase and mezzanines, profiled panels which is used as non-warmth-keeping walling as well as service loads on fire escape staircase and mezzanines. Safety factors for the design loads and effects as well as safety factor for the building responsibility were defined according to [12]. Live loads (or technological loads) were accepted according to the target specification. Calculation the design values for climate loadings has been performed according to the requirements [12]. So, tree types of snow loads and two types of wind load have been considered when optimum designing of the steel portal frames.

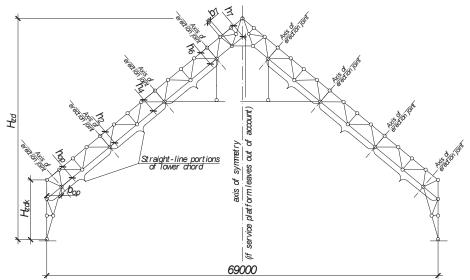
Design loads and effects have been combined in 16 design load case combinations taken into account the combination factors according to requirements of design code [12]. All loads and effects on the structure were presented as concentrated loads at the nodes and were determined analytically depending on the variable parameters of the geometrical scheme. Mathematical model of the parametric optimization task for the steel portal frame with lattice structural members has been formulated as nonlinear programming task including the set of design variables, system of constraints as well as specified purpose function.

3.1. Design variables

Parameters of the geometric scheme of the portal frames have been considered as design variables. Variable parameters of the geometrical scheme were building height at the eave node H_{zdk} and at the ridge H_{zd} , distance between upper and lower chords of the lattice rafter at the eave node hop and at the erection joints h_2 , h_4 , h_6 i h_7 , distance between chords of the lattice column at the eave node bop and parameter b_7 (see Fig. 3.2). Start values for the design variables were accepted according to the design decision of the steel framework developed by Open Join-Stock Company "V. Shymanovsky Ukrainian Research and Design Institute of Steel Construction", $H_{zd} = 39.58 \,\mathrm{m}, \qquad H_{zdk} = 10.63 \,\mathrm{m},$ namely: $h_2 = h_4 = h_6 = h_7 = h_{op} = b_{op} = 2.6 \,\mathrm{m}, \ b_7 = 0.56 \,\mathrm{m}.$ Additionally, cross-sectional sizes of the structural members with circle hollow sections (CHS) for each stiffness type were considered as design variables (see Table 3.1).

Table 3.1. Variable cross-sectional sizes for the CHS structural members of the portal frame

Destination and	Stiffness	Design variables	Start	Stiffness	Design variables	Start
location of	type	name, diameter×	values,	type	name, diameter×	values,
structural	number	thickness	mm×mm	number	thickness	mm×mm
member						
Chords of the lattice structural members	1	$d_1 \times t_1$	299×25	7	$d_6 \times t_7$	299×16
	2	$d_1 \times t_2$	299×14	8	$d_8 \times t_8$	299×10
	3	$d_1 \times t_3$	299×10	9	$d_9 \times t_9$	299×10
	4	$d_1 \times t_4$	299×14	10	$d_{10} \times t_{10}$	299×14
	5	$d_1 \times t_5$	299×14	11	$d_{11} \times t_{11}$	180×12
	6	$d_6 \times t_6$	299×10	_	_	_
Elements of the lattice rafters	12	$d_{12} \times t_{12}$	152×8	14	$d_{14} \times t_{14}$	102×5
	13	$d_{13} \times t_{13}$	121×8	_	_	_
Frame ridge	15	$d_{15} \times t_{15}$	152×8	17	$d_{17} \times t_{17}$	180×12
	16	$d_{16} \times t_{16}$	102×5	18	$d_{18} \times t_{18}$	180×12
Elements of the lattice columns	19	$d_{19} \times t_{19}$	299×10	21	$d_{21} \times t_{21}$	299×25
	20	$d_{19} \times t_{20}$	299×25	22	$d_{22} \times t_{22}$	102×5
Suspension arm of the service platform	23	$d_{23} \times t_{23}$	180×12	_	_	_



<u>Figure 3.2.</u> The design scheme of the portal frame with specification of the variable geometrical parameters

3.2. System of constraints

The system of constraints Eqs.(1.2) - (1.3)should cover strength and stability constraints formulated for all design sections of all structural members of the considered steel subjected to all structure design combinations at the ultimate limit state. The following strength constraints have included in the system of constraints Eqs. (1.2) - (1.3), formulated for all design sections, $\forall i = \overline{1, N}_{DS}$ (N_{DS} is the total number of the design sections in structural members), of all structural members, $\forall j = 1, N_B$ (N_B is the total number of the structural members), subjected to all ultimate load case combination, $\forall k = \overline{1, N_{LC}^{ULS}},$ namely normal stresses verifications:

$$\frac{N_{ijk}\left(\vec{X}\right)}{A_{n,j}\left(\vec{X}_{CS}\right)R_{y,j}\gamma_{c}}-1\leq 0;$$
(3.1)

where $\sigma_{ijk}(\vec{X}) = \frac{N_{ijk}(\vec{X})}{A_{n,j}(\vec{X}_{CS})}$ is the value of the

normal stresses caused by axial force $N_{iik}(\vec{X})$ acting in ith design section of jth structural member subjected to k th ultimate load case combination calculated from the equations system of the finite element method presented by Eq.(1.5); $A_{n,i}(\vec{X}_{CS})$ is the net cross-sectional area of jth structural member calculated depending on the variable crosssectional dimensions of the structural members \vec{X}_{CS} ; γ_c is the safety factor [12]; $R_{y,j}$ is the design strength for steel member subjected to tension, bending and compression; $R_{\nu}\gamma_{c}$ are allowable value for normal stresses [12]; $\sigma_{x,ijk}(\vec{X})$ are normal stresses at the specified cross-section point caused by internal forces acting in ith design section of jth structural member subjected to k th ultimate load case combination calculated from the linear equations system of the finite element method presented by Eq. (1.5). The value of the normal stresses $\sigma_{x,iik}(\vec{X})$ at the specified cross-section point has been calculated depending on the variable geometrical parameters of the structure

 \vec{X}_G and variable cross-sectional dimensions of the structural members \vec{X}_{CS} .

The following constraints on slenderness of the structural members have been included in the system of constraints Eqs. (1.2) – (1.3), $\forall j = \overline{1, N_B}$:

$$\frac{l_{ef,y,j}(\vec{X}_G)}{i_{y,j}(\vec{X}_{CS})\lambda_{uy,j}} - 1 \le 0;$$
 (3.2)

$$\frac{l_{ef,z,j}\left(\vec{X}_{G}\right)}{i_{z,j}\left(\vec{X}_{CS}\right)\lambda_{uz,j}}-1,0\leq0;$$
(3.3)

where $i_{y,j}(\vec{X}_{CS})$ and $i_{z,j}(\vec{X}_{CS})$ are radiuses of inertia for j th structural member's design cross-section relative to the main axis of inertia and calculated depending on the variable crosssectional dimensions of the structural members \vec{X}_{CS} ; $l_{ef,y,j}(\vec{X}_G)$ and $l_{ef,z,j}(\vec{X}_G)$ are design lengths for j th structural member in the main planes of inertia calculated depending on the variable geometrical parameters of the structure \vec{X}_G ; $\lambda_{uy,j}$ and $\lambda_{uz,j}$ are the ultimate slenderness for j th structural member. Design lengths of structural members $l_{ef,v,j}(\vec{X}_G)$ $l_{ef,z,j}(\vec{X}_G)$ were defined according to [12] as: for chords, support diagonals and support columns of the lattice structural members $l_{ef,y,j}(\vec{X}_G) = l_j(\vec{X}_G); \ l_{ef,z,j}(\vec{X}_G) = l_{1,j};$ for other elements of the lattice structural members $l_{ef,y,j}(\vec{X}_G) = 0.85 l_j(\vec{X}_G); \quad l_{ef,z,j}(\vec{X}_G) = 0.85 l_{1,j};$ here l_i is the geometrical length for j th bar of lattice structural member; $l_{1,j}$ is the distance between out-of-plane restraints of the member from the horizontal displacements in out-ofdirection. Ultimate values for slenderness of the lattice structural members specified according to [12] were

 $\lambda_{uy,j} = \lambda_{uz,j} = 400$ for all tensioned members; $\lambda_{uy,j} = \lambda_{uz,j} = 150$ for all compressed members.

The following stability constraints have been in the system of constraints included Eqs. (1.2) - (1.3), formulated for all design sections, $\forall i = \overline{1, N}_{DS}$, of the structural members subjected to all ultimate load case combination, $\forall k = \overline{1, N_{IC}^{ULS}},$ namely flexural buckling verifications for all structural members subjected to axial compression force $N_{ijk}(\vec{X})$, $\forall j = \overline{1, N_p}$:

$$\frac{N_{ijk}\left(\vec{X}\right)}{\varphi_{y,j}\left(\vec{X}_{G},\vec{X}_{CS}\right)A_{j}\left(\vec{X}_{CS}\right)R_{y,j}\gamma_{c}}-1\leq 0; \quad (3.4)$$

$$\frac{N_{ijk}\left(\vec{X}\right)}{\varphi_{z,j}\left(\vec{X}_{G}, \vec{X}_{CS}\right) A_{j}\left(\vec{X}_{CS}\right) R_{y,j} \gamma_{c}} - 1 \le 0; \quad (3.5)$$

where $A_j\left(\vec{X}_{CS}\right)$ is the gross cross-sectional area of j th structural member calculated depending on the variable cross-sectional dimensions of the structural members \vec{X}_{CS} ; $\varphi_{y,j}\left(\vec{X}_G,\vec{X}_{CS}\right)$ and $\varphi_{z,j}\left(\vec{X}_G,\vec{X}_{CS}\right)$ are column's stability factors corresponded to flexural buckling relative to the main axes of inertia and calculated depending on the design lengths $l_{ef,y,j}\left(\vec{X}_G\right)$, $l_{ef,z,j}\left(\vec{X}_G\right)$, cross-section type and cross-section geometrical properties for the j th structural member [12]. The flexural buckling factors $\varphi_{y,j}\left(\vec{X}_G,\vec{X}_{CS}\right)$ and $\varphi_{z,j}\left(\vec{X}_G,\vec{X}_{CS}\right)$ calculated depending on the variable geometrical parameters of the structure \vec{X}_G and variable cross-sectional dimensions of the structural members \vec{X}_{CS} .

The following local buckling constraints have been also included into the system of constraints:

$$\frac{\overline{\lambda}_{w,j}(\vec{X}_{CS})}{\overline{\lambda}_{uw,j}(\vec{X})} - 1 \le 0; \tag{3.6}$$

$$\frac{\overline{\lambda}_{f,j}(\vec{X}_{CS})}{\overline{\lambda}_{uf,j}(\vec{X})} - 1 \le 0$$
(3.7)

where $\overline{\lambda}_{w,i}(\vec{X}_{CS})$ and $\overline{\lambda}_{f,i}(\vec{X}_{CS})$ are the nondimensional slenderness of the web and flange respectively of the cross-section for structural member; $\overline{\lambda}_{uv,j}(\vec{X})$ and $\overline{\lambda}_{uf,j}(\vec{X})$ are the maximum values for corresponded nondimensional slenderness for column structural members calculated depending on the internal forces (ration of the bending moment to the axial force), as well as depending on the design lengths $l_{ef,y,j}$, $l_{ef,z,j}$, cross-section type and cross-section geometrical properties for the j th structural member [12]. The non-dimensional $\overline{\lambda}_{w,i}(\vec{X}_{CS})$ and $\overline{\lambda}_{f,j}(\vec{X}_{CS})$ slenderness calculated depending on the variable crosssectional dimensions of the structural members \vec{X}_{CS} only. At the same time, the maximum values for corresponded non-dimensional slenderness $\overline{\lambda}_{uw,i}(\vec{X})$ and $\overline{\lambda}_{uf,i}(\vec{X})$ calculated depending on the variable geometrical parameters of the structure \vec{X}_G and variable cross-sectional dimensions of the structural members \vec{X}_{CS} .

The system of constraints Eqs. (1.2) - (1.3)has been also covered the displacements constraints (stiffness inequalities) for the specified nodes of the considered steel structure subjected to all design combinations at the serviceability limit state. The following horizontal and vertical displacements constraints have been included into the system of constraints Eqs. (1.2) -(1.3), formulated for all nodes, $\forall m = \overline{1, N_N}$ (N_N) is the total number of nodes in the considered steel structure), of the steel

structure subjected to all serviceability load case combination, $\forall k = \overline{1, N_{LC}^{SLS}}$, namely:

$$\frac{\delta_{x,mk}\left(\vec{X}\right)}{\delta_{yx,m}} - 1 \le 0; \tag{3.8}$$

$$\frac{\delta_{z,mk}\left(\vec{X}\right)}{\delta_{yz,m}} - 1 \le 0; \tag{3.9}$$

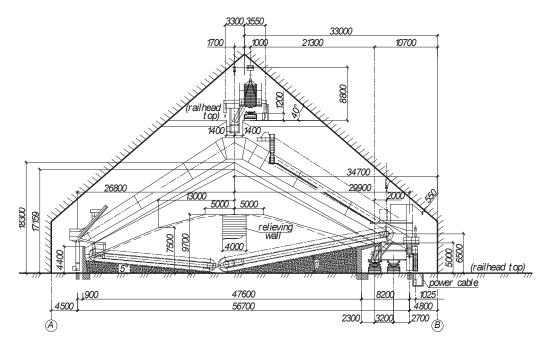
where $\delta_{x,mk}\left(\vec{X}\right)$ and $\delta_{z,mk}\left(\vec{X}\right)$ are the horizontal and vertical displacements respectively for l th node of the steel structure subjected to k th serviceability load case combination calculated from the linear equations system of the finite element method presented by Eq. (1.6); $\delta_{ux,m}$ and $\delta_{uz,m}$ are the allowable horizontal and vertical displacements for m th structural node. Ultimate values for linear node displacements of the steel lattice portal frame were calculated according to [12] as $\delta_{ux,m} = H_{zd}/210$ and $\delta_{uz,m} = L/300 = 230$ mm.

Additional requirements that describe structural, technological and serviceability particularities of the considered structure, as well as constraints on the building functional volume (see Fig. 3.3) can be also included into the system Eqs. (1.2) - (1.3). In particular these requirements can be presented in the form of constraints on lower and upper values of the design variables, $\forall t = \overline{1, N_x}$:

$$1 - \frac{X_{\iota}}{X_{\iota}^{L}} \le 0 \; ; \tag{3.10}$$

$$\frac{X_{t}}{X_{t}^{U}} - 1 \le 0; (3.11)$$

where X_{ι}^{L} and X_{ι}^{U} are the lower and upper bounds for the design variable X_{ι} ; N_{X} is the total number of the design variables.



<u>Figure 3.3.</u> Technological equipments and constraints that describe useful space in the building

Additional constraints on cross-sectional sizes of the structural members with circle hollow sections type have been formulated according to the requirements of [12]. There were constraints permissible minimal thickness permissible maximum diameter-to-thickness ratio for the structural members, namely, $\forall l = 1, N_{ST}$:

$$1,0 - \frac{t_l}{t_{\min l}} \le 0; (3.12)$$

$$1, 0 - \frac{t_l}{t_{\min,l}} \le 0;$$

$$\frac{D_l}{t_l \delta_{\max,l}} - 1, 0 \le 0;$$
(3.12)

where l is the number of the stiffness type; N_{ST} is the overall quantity of the stiffness types in the considered steel lattice portal frame; t_i and D_{i} are thickness and diameter of the circle hollow section for 1th stiffness type (see Table 3.1) respectively; $t_{min,l}$ is the minimum thickness of the circle hollow section in accordance with design code [12] $t_{\min,l} = 3 \,\mathrm{mm}$ for chords, support diagonals and support columns of lattice structural members and $t_{\min,l} = 2.5 \,\text{mm}$ for other lattice elements; $\delta_{\text{max} I}$ is the maximum diameter-to-thickness ratio for the structural member with circle hollow section in accordance with design code [12] depending on the yield stress value $R_{vn} = 245 \,\text{MPa} < 295 \,\text{MPa}$ as $\delta_{\text{max},l} = 30$ for chords elements of the lattice structural members and $\delta_{\text{max},l} = 90$ for other lattice elements.

The following constraints that describe conditions for designing gusset-less welded between CHS structural members formulated according to the requirements [12] have been also included in the system Eqs. (1.2) - (1.3), $q, p = 1...N_{ST}$:

$$0,3d_p \le d_q \le d_p;$$
 (3.14)

where p and q are the numbers of the stiffness types of structural members connected in the joint, here p is the number of the chord's stiffness type; q is the number of the lattice stiffness type.

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3.3. Objective function

Minimum weight as well as minimum construction budget has been considered as purpose function. Analytical expression for the structural weight depending on the variable cross-sectional sizes of the members can be written by the following formula:

$$\mathbf{M}(\vec{X}_{G}, \vec{X}_{CS}) = \varphi \rho \sum_{j=1}^{N_{E}} A_{j} l_{j} =$$

$$= 4\pi \varphi \rho \sum_{j=1}^{N_{E}} t_{j} (D_{j} - t_{j}) l_{j} \rightarrow \min;$$
(3.15)

where A_j and l_j are gross cross-sectional area and geometrical length for jth structural member respectively; ρ – steel density, $\rho = 7850\,\mathrm{kg/m3}$; φ is the factor that takes into account the increment of structural weight due to the present of the adjunct elements in the structural members and joints (stiffeners, ribs, end-plates, gussets etc.), φ was defined according to the steel specification mentioned in the source project for the warehouse framework, $\varphi \square 1,1$.

Construction budget of the steel portal frame with lattice structural members taken into account construction budget of mezzanines erected at the level +28,25 can be presented as follow:

$$\mathbf{K} = C_{mnf}^{lt} + C_{mnf}^{sl} + C_{asm} + C_{asm}^{wl} + \\ + C_{c.p.} + C_{a.c.} + C_{mat};$$

where C_{mnf}^{lt} is the manufacturing cost of lattice structural members; C_{mnf}^{sl} is the manufacturing cost of mezzanine's structural members; C_{asm} is the assembly cost for steel portal frame; C_{asm}^{wl} is the assembly cost for walling; $C_{c.p.}$ is the cost on the work package for corrosion protection of the steel framework; $C_{q.c.}$ is the cost for the quality control of welded connections; C_{mat} is

the material cost for structural members of the steel portal frame. Analytical expression of the construction budget for manufacturing and erection of the steel lattice portal frame depending on the design variables have been presented by the following, UAH:

$$K = 20670 \mathbf{M} \left(\vec{X}_{G}, \vec{X}_{CS} \right) + 1525 D_{ov} \left(\vec{X}_{G} \right) +$$

$$+1300 H_{zdk} + 110 \mathbf{A}_{s} \left(\vec{X}_{G}, \vec{X}_{CS} \right) +$$

$$+21120 \rightarrow \min;$$
(3.16)

where D_{ov} is the half-rafter length, $D_{ov} = \left(\left(0,5L \right)^2 + \left(H_{zd} - H_{zdk} \right)^2 \right)^{\frac{1}{2}};$ L is the portal frame span, $L = 69 \, \mathrm{m};$ $\mathbf{A_s} \left(\vec{X}_G, \vec{X}_{CS} \right)$ is the total surface area of the steel lattice portal frame to be subjected to anti-corrosion treatment.

4. PARAMETRIC OPTIMIZATION ALGORITHM BASED ON THE GRADIENT PROJECTION METHOD

Let present the following numerical algorithm to solve the parametric optimization problem for steel structures formulated above.

Step 1. Describing an initial design (a set of design variables) and initial data for structural optimization.

The design variable vector $\vec{X}_k = \left(\vec{X}_G, \vec{X}_{CS}\right)_k^T$ has been specified, where k is the iteration index, k=0. The structural topology, cross-section types and node type connections of the bars, the support conditions of the bar system, as well as loading patterns, load case combinations and load design values are prescribed and constants.

Initial data for optimization of the considered steel structure are design strength for steel member R_y , safety factor γ_c , factors to define flexural design lengths $l_{ef,y,j}$, $l_{ef,z,j}$ for all column structural members; allowable values

for horizontal and vertical displacements $\delta_{ux,l}$ and $\delta_{uz,l}$ of the specified nodes of the considered steel structure; lower \vec{X}^L and upper \vec{X}^U bounds for the design variables; as well as specified objective function $f(\vec{X}_k)$.

Step 2. Calculation of the geometrical and design lengths for all structural members.

The geometrical lengths l_i of all structural members are calculated based on the node coordinates of the considered steel structure. The latter depend on the unknown (variable) geometrical parameters of the structure X_G . The design lengths $l_{\mathit{ef},\mathit{y},\mathit{j}}$, $l_{\mathit{ef},\mathit{z},\mathit{j}}$ of all column structural members are calculated using calculated geometrical lengths l_i and initial data relating to the design length factors. The latter are constant during the iteration process presented below. Variation of the geometrical lengths l_i and corresponded design lengths $l_{ef,y,j}$, $l_{ef,z,j}$ on the further iterations has been performed based on the current values of the variable (unknown) parameters X_G of the geometrical scheme.

Step 3. Calculation of the cross-section dimensions and geometrical properties for all design cross-sections.

Geometrical properties of the design cross-sections (areas, moments of inertia, elastic section moments, radiuses of inertia, etc.), as well as non-dimensional slenderness for cross-section elements (webs and flanges) $\overline{\lambda}_{w,j}\left(\vec{X}_{CS}\right)$

and $\bar{\lambda}_{f,j}(\bar{X}_{CS})$ have been calculated depending on the current values of the unknown (variable) cross-section dimensions \bar{X}_{CS} .

Step 4. Linear structural analysis of the considered steel structure.

For each m th node of the finite element model subjected to k th serviceability load case combination the displacements and rotations, as well as the design horizontal $\delta_{x,mk}(\vec{X})$ and

vertical $\delta_{z,lk}(\vec{X})$ displacements can be calculated using the linear equations system of the finite element method presented by Eq. (1.6).

For each i th design section of j th structural member subjected to k th ultimate load case combination the design internal forces can be calculated using the linear equations system of the finite element method presented by Eq. (1.5).

Step 5. Calculation of the state variables (stresses, buckling factors, allowable non-dimensional slenderness etc.).

The value of the normal $\sigma_{x,ijk}(\vec{X})$ stresses at the specified cross-section point has been calculated depending on the axial force acting in i th design section of j th structural member subjected to k th ultimate load case combination as presented by the design code.

The flexural buckling factors $\varphi_{y,j}(\vec{X}_G, \vec{X}_{CS})$,

 $\varphi_{z,j}\left(\vec{X}_G, \vec{X}_{CS}\right)$ have been calculated depending on the corresponded design lengths, cross-section type and cross-section geometrical properties for the structural members according to the design code [12].

The maximum values for corresponded non-dimensional slenderness $\overline{\lambda}_{uw,j}(\vec{X})$ and

 $\overline{\lambda}_{uf,j}(\vec{X})$ for column structural members have been calculated depending on the design lengths $l_{ef,y,j}$, $l_{ef,z,j}$, cross-section type and cross-section geometrical properties for the jth structural member [12].

Step 6. Verifications of the constraints and construction the set of active constraints numbers A.

Verification of the constraints Eqs. (3.1), (3.4), (3.5) has been performed for all ultimate load case combinations and all design cross-sections of all structural members. Verification of the constraints Eqs. (3.8), (3.9) have been also conducted for all serviceability load case combinations and all design structural nodes.

Additional requirements in the form of constraints Eqs. (3.10), (3.11) on lower and upper values of the design variables, local constraints Eqs. (3.6), buckling (3.7),constraints on the member's slenderness (3.3), constraint Eq. (3.12) on Eqs. (3.2), permissible minimal thickness, constraint Eq. (3.13) on permissible maximum diameterto-thickness ratio for the structural members, as well as the conditions Eq. (3.14) for designing gusset-less welded joints between structural members with circle hollow sections have been also verified. Set of the active constraints numbers calculated for the current approximation X_k has been constructed according to Eq. (2.2).

Step 7. Calculation of the current objective function value $f(\vec{X}_k)$, objective function gradient $\nabla f(\vec{X}_k)$ and determination of the desired decrement of the objective function value $\Delta f(\vec{X}_k)$.

The objective function gradient $\nabla f(\vec{X}_k)$ can be calculated by the numerical differentiation with respect to the design variables using the finite difference approximation. The desired decrement of the objective function value $\Delta f(\vec{X}_k)$ can be assigned as 5...25% from the current objective function value $f(\vec{X}_k)$.

Step 8. Construction of the constraint's violations vector \mathbf{V} and the matrix of the active constraint's gradients $[\nabla \varphi]$. The vector of the values of the constraint's violations \mathbf{V} and the matrix of the constraint's gradients $[\nabla \varphi]$ are constructed for active constraints only according to the set of active constraints numbers \mathbf{A} .

Step 9. Construction the matrix of active linearindependent constraint's gradients with triangular structure. The set of independent constraint's numbers L and the matrix of active linear-independent constraint's gradients $\mathbf{H}[\nabla \varphi]$ with triangular structure are constructed according to the algorithm presented by the paper [14].

Step 10. Step parameter ξ calculation. Step parameter ξ has been calculated according to Eq. (2.11) or Eq. (2.12) and can be modified on the further iterations depending on convergence of the iterative process presented by Eq. (2.1). Step 11. Calculation the column-vectors $\vec{\mu}_{\perp}$ and $\vec{\mu}_{\square}$ which define the design variables increment subject to the condition of elimination the

improvement of the objective function value. The vectors $\vec{\mu}_{\perp}$ and $\vec{\mu}_{\square}$ can be calculated using Eq. (2.18) and Eq. (2.19) respectively.

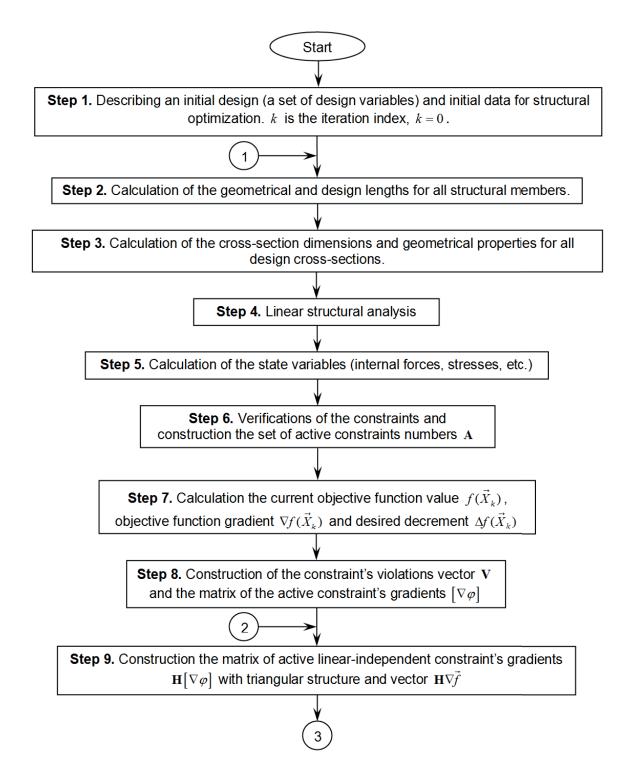
constraint's violations and subject to the

If some h th component of the column-vectors $\vec{\mu}_{\perp}$ and $\vec{\mu}_{\square}$ satisfies Eq. (2.17), the corresponded constraint gradient $\nabla \varphi_h$ has been excluded from the matrix $[\nabla \varphi]$, and corresponded violations V_h has been excluded from the vector \mathbf{V} , as well as the return to step 9 has to be conducted. In contrary case transition to the step 11 has been performed.

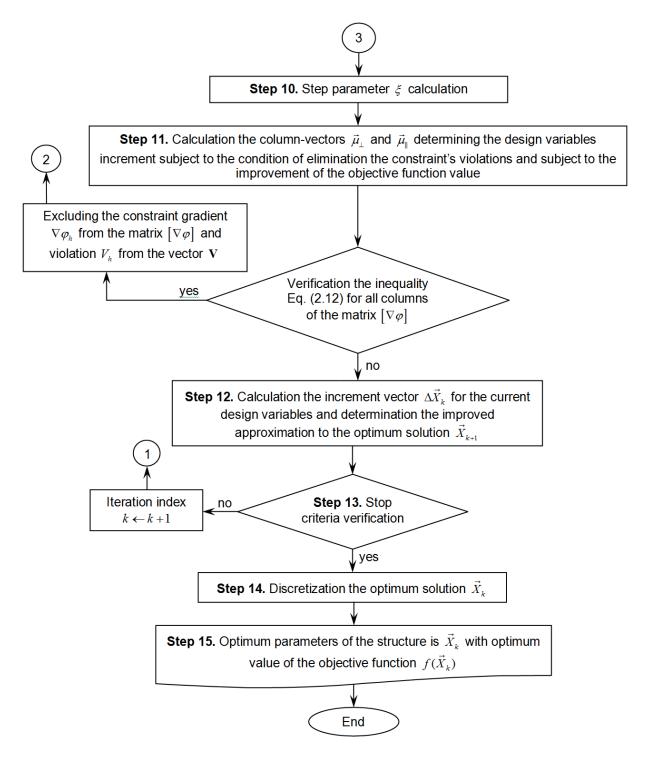
Step 12. Calculation the increment vector for the current design variables and determination the improved approximation to the optimum solution. The increment vector $\Delta \vec{X}_k$ for the current design variables values \vec{X}_k has been calculated according to Eq. (2.20) or Eq. (2.21). The improved approximation \vec{X}_{k+1} to the optimum solution has been determined according to Eq. (2.1).

Step 13. Stop criteria verification of iterative searching for the optimum solution. If all constraints Eqs. (3.1) - (3.14) are satisfied with appropriate accuracy, as well as inequality Eq. (2.22) or one of the stop criteria described by the paper [14] is also satisfied, then transition to the step 13 has been performed. In contrary case return to the step 1 has been conducted with $k \leftarrow k+1$.

Step 14. Discretization the optimum solution \vec{X}_k obtained in the continuum space of the design variables.



<u>Figure 4.1.</u> The flow chart for structural optimization according to the searching technique based on the gradient projection method



<u>Figure 4.1.</u> (continuation). The flow chart for structural optimization according to the searching technique based on the gradient projection method

Step 15. Optimum parameters of the structure is \vec{X}_k with optimum value of the objective

function $f(\vec{X}_k)$.

Figure 4.1 presents the flow chart for structural optimization according to the searching

technique describing by the gradient projection method considered above.

5. OPTIMIZATION RESULTS

A parametric optimization methodology presented above has been realized in software OptCAD [19]. This software provides solutions to a wide range of problems, namely: (i) linear static analysis of bar structures; (ii) verification of the load-bearing capacity of the structural members according to specified design code; (iii) searching for

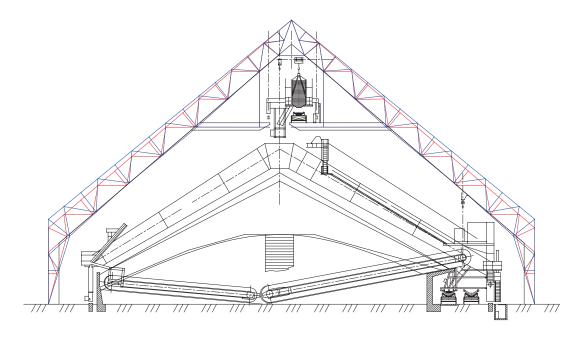
values of the structural parameters when complies with structure design code requirements and designer's criterions; (iv) parametric optimization of the steel bar structures by the determined criterion. Mathematical apparatus of the software combines the finite element method to perform linear static analysis of the bar system, as well as improved gradient projection method to solve parametric optimization problems formulated as nonlinear programming tasks [19].

<u>Table 5.1.</u> Optimal values for variable geometrical parameters of the portal frame's design scheme

Design variable		Optimum values, m, by the criterion of minimum				
	m	weight		costs on fabrication and erection		
			when lower cho	ord of lattice rafter is		
		straight-line	polygonal	straight-line	polygonal	
H_{zd}	39.58	38.82	39.55	38.74	39.44	
H_{zdk}	10.63	11.92	11.84	12.08	12.03	
h_2	2.60	_	3.23	_	3.31	
h_4	2.60	_	3.21	_	3.23	
h_6	2.60	_	2.68	_	2.70	
h_{7}	2.60	2.18	2.68	2.13	2.61	
b_7	0.55773	1.00	0.56	1.05	0.63	
h_{op}	2.60	3.61	3.72	3.73	3.87	
b_{op}	2.60	3.08	3.02	3.02	2.95	
Weight, ×10³,kg	30.78	18.98	18.45			
Costs, UAH	786681			534647	524257	

<u>Table 5.2</u>. Optimal values for variable cross-sectional sizes of the lattice structural members of the portal frame

	<u> </u>	Ontir	num voluos	mm×mm	by the criteria	on of minim	ulim	
		Optimum values, mm×mm, by the criterion of minimum weight costs on fabrication and erection						
Design variable	Start value, m	when lower chord of the lattice rafter is						
		straight-line			straight-line		gonal	
		in the search space						
		continuous		†	discrete contin		discrete	
$d_1 \times t_1$	299×25	183.4×8.4	192.3×8.6	194×9.0	173.9×9.3		168×11.0	
$d_1 \times t_2$	299×14	183.4×6.1	192.3×6.4	194×7.0	173.9×6.9	172.9×6.3	168×8.0	
$d_1 \times t_3$	299×10	183.4×6.9	192.3×6.5	194×6.5	173.9×7.3	172.9×7.4	168×8.5	
$d_1 \times t_4$	299×14	183.4×8.5	192.3×9.3	194×9.0	173.9×10.7	172.9×9.3	168×11.0	
$d_1 \times t_5$	299×14	183.4×6.1	192.3×6.4	194×6.5	173.9×5.8	172.9×5.8	168×6.0	
$d_6 \times t_6$	299×10	297.4×9.9	303.1×10.1	299×10.0	298.1×9.9	292.3×9.7	299×10.0	
$d_6 \times t_7$	299×16	297.4×11.3	303.1×11.4	299×12.0	298.1×11.6	292.3×11.	299×12.0	
$d_8 \times t_8$	299×10	264.4×8.8	269.4×9.0	299×10.0	269.0×9.0	263.6×8.8	273×9.5	
$d_9 \times t_9$	299×10	203.0×6.8	210.7×7.0	194×8.5	211.5×7.0	202.8×6.8	219×7.5	
$d_{10} \times t_{10}$	299×14	143.5×4.8	147.1×4.9	152×5.5	148.1×4.9	143.3×4.8	146×5.5	
$d_{11} \times t_{11}$	180×12	183.4×9.7	192.3×9.6	194×9.5	173.9×11.5	172.9×11.	168×13.0	
$d_{12} \times t_{12}$	152×8	161.2×4.5	165.6×4.6	152×5.5	164.9×4.6	160.8×4.5	168×5.0	
$d_{13} \times t_{13}$	121×8	110.3×3.9	107.4×3.6	108×4.0	80.7×4.9	79.1×5.6	83×5.0	
$d_{14} \times t_{14}$	102×5	89.2×3.5	90.9×3.5	95×3.5	89.4×3.5	87.7×3.5	95×3.5	
$d_{15} \times t_{15}$	152×8	143.5×5.2	147.1×5.1	152×5.0	133.5×5.9	135.1×5.6	133×6.0	
$d_{16} \times t_{16}$	102×5	55.0×3.5	57.9×3.5	60×3.5	52.2×3.5	51.9×3.5	54×3.5	
$d_{17} \times t_{17}$	180×12	112.8×4.9	116.3×4.7	108×5.0	90.4×6.1	101.8×5.4	95×6.0	
$d_{18} \times t_{18}$	180×12	143.5×6.3	147.1×6.6	152×6.5	148.1×6.6	143.3×6.4	146×7.5	
$d_{19} \times t_{19}$	299×10	297.4×9.9	303.2×10.1	299×10.0	298.1×9.9	292.3×9.7	299×10.0	
$d_{19} \times t_{20}$	299×25	297.4×9.9	303.2×10.1	299×10.0	298.1×9.9	292.3×9.7	299×10.0	
$d_{21} \times t_{21}$	299×25	297.4×21.0	303.2×21.1	299×22.0	298.1×22.3	292.3×22.	299×24.0	
$d_{22} \times t_{22}$	102×5	89.2×3.5	90.9×3.5	95×3.5	89.4×3.5	87.7×3.5	95×3.5	
$d_{23} \times t_{23}$	180×12	148.9×5.3	151.6×5.3	152×5.5	79.7×11.4	81.8×10.6	83×1.2	
Weight, ×10 ³ ,kg	30.78	18.45	18.98	19.70	_		_	
Costs, UAH	786681		_	_	534647	524257	552368	



<u>Figure 5.1.</u> The optimum design decision by the criterion of minimum structural weight when lower chord of the lattice rafter is straight-line

Formulated parametric optimization problem for the steel lattice portal frame has been solved using software OptCAD. Task dimensions are: account of design variables is 50, account of problem constraints is 14000.

Optimization results received using software OptCAD are presented by the Tables 5.1 and 5.2. Figure 5.1 show optimal design decision by the criterion of minimum structural weight of the steel lattice portal frame (project with start values of the design variables is indicated by the red color, optimum project is indicated by the blue color).

CONCLUSIONS

The results of the presented study can be formulated as follow:

1. The paper has proposed a mathematical model for parametric optimization problem of the steel lattice portal frame with CHS structural members. The design variable vector includes geometrical parameters of the structure (node coordinates), as well as cross-sectional

dimensions of the structural members. The system of constraints covers load-carrying capacities constraints formulated for all design sections of structural members of the steel structure subjected to all ultimate load case combinations. The displacements constraints formulated for the specified nodes of the steel structure subjected to all serviceability load case combinations have been also included into the system of constraints. Additional requirements in the form of constraints on lower and upper values of the design variables, constraints on permissible minimal thicknesses, constraints on permissible maximum diameter-to-thickness ratio for the structural members with circle hollow sections, as well as the conditions for designing gusset-less welded joints between structural members with circle hollow sections have been also considered in the scope of the mathematical model.

2. The method of the objective function gradient projection onto the active constraints surface with simultaneous correction of the constraints violations has been applied to solve the formulated parametric optimization problem.

- 3. A numerical algorithm for solving the parametric optimization problems of steel lattice portal frames with CHS structural members has been presented in the paper.
- 4. New optimal layouts of the steel lattice portal frame by the criterion of the minimum weight, as well as minimum costs on manufacturing and erection have been shown.

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